Falcon Hypersonic Technology Vehicle (HTV-2)

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Program Objectives

Develop and test an unmanned, rocket-launched, maneuverable, hypersonic air vehicle that glides through the Earth's atmosphere up to Mach 20 speed.

Performance Metrics

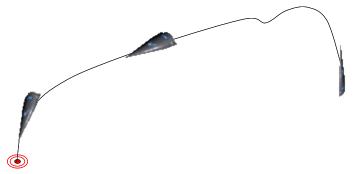
- High lift-to-drag ratio, advanced carbon-carbon aeroshell.
- High-temperature materials and structures.
- Precision navigation, guidance, and control.
- Proper function of onboard sensors and off-board auxiliary measurements to characterize:
 - Vehicle state.
 - Flight environment.
 - Aerodynamic and aerothermal performance.
 - GNC system performance.

Program Status

- Flight #1 completed 22 April 2010:
 - Preliminary review of data indicates the HTV-2 achieved controlled flight within the atmosphere at over Mach 20.
 - Lost vehicle telemetry approximately 9 minutes into the mission.
- Flight #2 planned for 10-17 Aug 2011.









Inaugural launch of the Minotaur IV booster

• Demonstration of the ability to fly at extreme angles of attack up to 89 degrees to meet stringent release requirements for the HTV-2.

Extensive data captured for critical areas of interest

- Aerothermal, aerodynamic, thermal protection, navigation, guidance, and control in the hypersonic flight regime.
- Successful demonstration of the first ever use of an autonomous flight termination system.

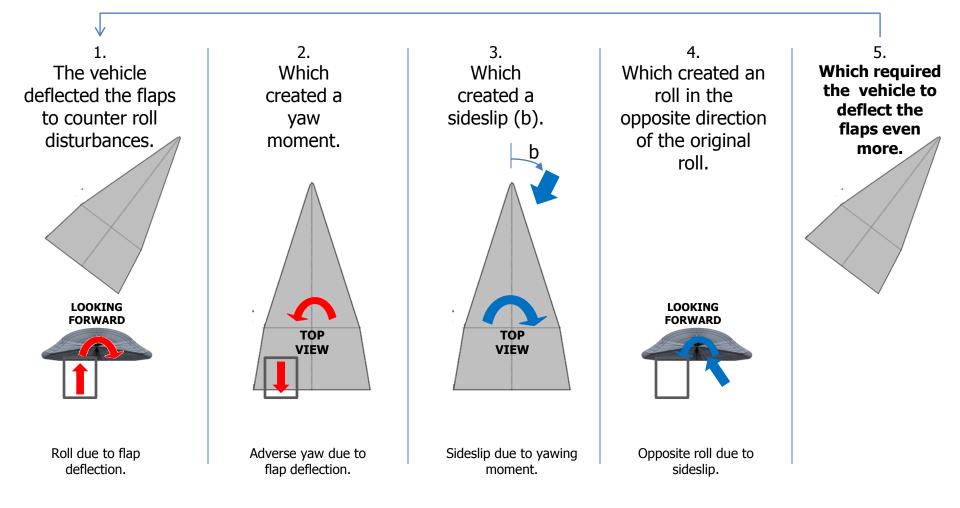
Flight dynamic anomaly observed during pull-up phase of flight

"Higher than predicted adverse yaw coupled into roll that exceeded the available roll control capability."

(Paulson, 2010, 1)



Roll-yaw coupling



6.) Eventually, the adverse roll moment from the sideslip and disturbances overwhelmed the roll moment available from the differential flap deflections.

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Independent Engineering Review Board (ERB)

- ERB chartered to examine data collected during flight #1.
- Extensive six-month review concluded the week of 13 Sep 2010.
- Probable cause and contributing factors to flight anomaly identified.

Key findings

- Higher-than-predicted yaw, which coupled into roll thus exceeding the available control capability at the time of the anomaly.
- This resulted in flight control authority limitations to operate at the angle of attack the vehicle was programmed to fly for the speed and altitude where the anomaly occurred.

Remediation plan

- Decrease the angle of attack.
- Adjust the vehicle's center of gravity.
- Use onboard reaction control to augment vehicle flaps.
- Can be implemented via software, ballast and flight profile changes.



ERB conclusion

- Probable cause and contributing factors identified.
- Closure of anomaly investigation phase.
- Recommendation of implementation of remediation actions.
- Effectiveness of remediation strategy to be reviewed at MDR.

Flight test report to be completed on 15 Jan 2011.

Remediation actions

- Engineering fixes (CG, AoA, RCS).
- Scientific fixes (aero model, autopilot, controls).
- Systems engineering and process.

No major redesign needed for flight #2

- High confidence in schedule completion.
- Flight #2 planned for 10-17 Aug 2011.



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